

UNCLASSIFIED

AD 295 002

*Reproduced
by the*

**ARMED SERVICES TECHNICAL INFORMATION AGENCY
ARLINGTON HALL STATION
ARLINGTON 12, VIRGINIA**



UNCLASSIFIED

NOTICE: When government or other drawings, specifications or other data are used for any purpose other than in connection with a definitely related government procurement operation, the U. S. Government thereby incurs no responsibility, nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use or sell any patented invention that may in any way be related thereto.

295002

ASTIA

AS 41 110

**Columbia University
in the City of New York**

**DEPARTMENT OF CIVIL ENGINEERING
AND ENGINEERING MECHANICS**



**INERTIA TERMS ASSOCIATED WITH SMALL MOTIONS
OF A SUBMERGED FLEXIBLE RECTANGULAR CYLINDER**

by

Jacob Lubliner and Hans H. Bleich

**Office of Naval Research
Project NR 064-428
Contract Nonr 266(86)
Technical Report No. 30
CU-1-62 ONR-266(86) - CE**

November 1962

**Reproduction in whole or in part is permitted for any purpose
of the United States Government.**

**Columbia University
in the City of New York**

**DEPARTMENT OF CIVIL ENGINEERING
AND ENGINEERING MECHANICS**



**INERTIA TERMS ASSOCIATED WITH SMALL MOTIONS
OF A SUBMERGED FLEXIBLE RECTANGULAR CYLINDER**

by

Jacob Lubliner and Hans H. Bleich

**Office of Naval Research
Project NR 064-428
Contract Nonr 266(86)
Technical Report No. 30
CU-1-62 ONR-266(86) - CE**

November 1962

**Reproduction in whole or in part is permitted for any purpose
of the United States Government.**

ABSTRACT

An expression for the kinetic energy of an infinite liquid mass due to small motions of a submerged cylinder is obtained analytically. When the motion is described in terms of generalized coordinates, the corresponding inertia coefficients appear in the kinetic-energy expression. The analysis is carried out for the problem of a floating flexible rectangular box. Calculations are performed for generalized coordinates representing rigid-body motion and plastic deformation.

I. Introduction

The purpose of this paper is to provide the background for the treatment of transient problems of elastic or plastic box structures resembling a surface ship and floating on the surface of a semi-infinite fluid. Whenever the transient problem is such that the fluid may be considered incompressible, the problem can be formulated in terms of generalized coordinates; the presence of the fluid then produces only terms which can be derived from the kinetic energy of the fluid, expressed in the generalized velocities. The results of this paper are intended to be applied to the determination of the elastic or plastic response of a ship-like structure to initial velocities imparted to it by an underwater explosion. Appropriate velocity distributions may be obtained by experimental or analytical means; an analytical approach has been presented in Ref. 1.

The present problem becomes amenable to analysis if it is treated as a two-dimensional one, that is, if the box is regarded as an infinitely long, flexible rectangular cylinder whose boundary undergoes small motions perpendicular to the generatrices; the latter remain straight and parallel to the axis. The fluid is assumed at rest at large distances from the cylinder, and the motions of the cylindrical boundary are small deviations from its rest configuration.

The problem may thus be viewed, approximately, as a boundary-value problem in plane ideal-fluid flow with prescribed normal velocities along a stationary boundary. It can therefore be attacked by the classical methods of plane potential theory, and, in particular, by the conformal mapping of the given boundary curve into a circle, and subsequent contour integration. The analysis is carried out for the general case in Section II, resulting in the general expression for the inertia coefficients. The expression takes the form of an infinite series whose terms depend on the Fourier coefficients of the transformed normal-velocity distribution. These Fourier coefficients may be obtained by quadrature (numerical, in general), provided the mapping function is known. The mapping function appropriate to the present case of a floating rectangular box is developed analytically in Section III, and is tabulated at the end of the report in a form suitable for integration, for the two beam-draft ratios of 2 and 3.

Calculations were actually carried out for the special case of motion in which the degrees of freedom considered were (i) the three rigid-body motions (horizontal and vertical translations, and rotation), and (ii) three deformations of the box representing possible plastic damage. These degrees of freedom are illustrated in Figure 1. The results are tabulated in the form of inertia matrices for the two beam-draft ratios considered.

An idea of the accuracy of the calculations (which, besides numerical integrations, involve truncations of infinite series) may be had by virtue of the fact that for the two translatory degrees of freedom the inertia coefficients may be calculated exactly. This is shown in the Appendix.

The results of this paper are also applicable to vibration problems, provided the frequencies are low enough to permit the assumption of incompressibility.

II. Virtual Inertia Terms for a Closed Cylinder

The kinetic energy of an incompressible fluid in irrotational two-dimensional flow in the $x y$ plane is, per unit distance in the z direction,

$$T = \frac{1}{2} \rho \oint \phi \, d\psi , \quad (1)$$

Where ρ is the (constant) fluid density, ϕ the velocity potential, ψ the stream function, and the integration is carried out in the positive direction around the boundary. (See, for example, Ref. 2, p. 66.) If the fluid is bounded internally by a closed cylinder, then, in accordance with the usual convention, the direction of integration is clockwise around the projection C of the cylinder. On introducing the complex potential function,

$$w = \phi + i\psi ,$$

one may write Equation (1) as

$$T = \frac{1}{4i} \rho \oint w^* dw , \quad (2)$$

(the asterisk denotes the complex conjugate), since

$$w^* dw = (\phi - i\psi)(d\phi + id\psi) = 2i\phi d\psi + \frac{1}{2} d(\phi^2 + \psi^2 - 2i\phi\psi),$$

and the integral of a perfect differential around a closed curve is zero.

The x and y components of fluid velocity, u and v, are given by the relation

$$u - iv = dw/dz , \quad (3)$$

where

$$z = x + iy .$$

The normal velocity (considered positive into the fluid) at a point $z = z_c$ on the curve C is therefore

$$\begin{aligned} u_v &= u \frac{dy}{ds} - v \frac{dx}{ds} \\ &= \text{Im}(dw/ds)_{z=z_c} \end{aligned} \quad (4)$$

where ds ($= |dz|$) is an element of arc length on C , measured positive counterclockwise.

If the z plane is now mapped conformally into a ζ plane such that the outside of C is mapped into the outside of the unit circle,

$$\zeta = e^{i\theta} ,$$

then for a fluid which is at rest at infinity the complex potential function may be expressed, in general, in the form

$$w = \sum_{n=1}^{\infty} \frac{a_n + ib_n}{n} \zeta^{-n} , \quad (5)$$

where the coefficients a_n and b_n are real. The normal velocity on C is now given by

$$u_v = \frac{d\theta}{ds} \sum_{n=1}^{\infty} (a_n \cos n\theta + b_n \sin n\theta) . \quad (6)$$

Consequently, by Fourier's theorem,

$$\begin{Bmatrix} a_n \\ b_n \end{Bmatrix} = \frac{1}{\pi} \int_0^{2\pi} u_v \begin{Bmatrix} \cos n\theta \\ \sin n\theta \end{Bmatrix} \frac{ds}{d\theta} d\theta . \quad (7)$$

The arc length on C , measured counterclockwise from the point corresponding to $\theta = 0$, may be treated as a function of θ , defined by

$$s(\theta) = \int_0^{\theta} \frac{ds}{d\theta} d\theta , \quad (8)$$

where

$$\frac{ds}{d\theta} = \left| \frac{dz}{d\zeta} \right| \quad \zeta = \exp(i\theta) . \quad (9)$$

In particular,

$$s(2\pi) \equiv p ,$$

the perimeter of C . We may further define a dimensionless variable,

$$\sigma \equiv s/p ,$$

and consider θ a function of σ . Equation (7) may then be written alternatively as

$$\begin{Bmatrix} a_n \\ b_n \end{Bmatrix} = \frac{p}{\pi} \int_0^1 u_v(\sigma) \begin{Bmatrix} \cos n\theta(\sigma) \\ \sin n\theta(\sigma) \end{Bmatrix} d\sigma . \quad (10)$$

When the expression (5) for the complex potential function is used in Equation (2) and the integration is carried out along $\zeta = \exp(i\theta)$; the resulting expression for the kinetic energy is

$$T = \frac{1}{2} \pi \rho \sum_{n=1}^{\infty} \frac{a_n^2 + b_n^2}{n} . \quad (11)$$

A small normal displacement of C may in general be regarded as the superposition of normal displacements corresponding to different modes of motion (or degrees of freedom), each of which may be expressed, at any time t , as

$$f_i(\sigma)q_i(t) ,$$

where $f_i(\sigma)$ describes the shape, while $q_i(t)$ gives the amplitude. (It should be noted that if q_i is a length, f_i is dimensionless, while if q_i is an angle, f_i has the dimension of length.) The general normal velocity is consequently

$$u_v = \sum_i f_i(\sigma) \dot{q}_i . \quad (12)$$

The Fourier coefficients a_n, b_n , are then

$$\begin{Bmatrix} a_n \\ b_n \end{Bmatrix} = \frac{p}{\pi} \sum_i \begin{Bmatrix} c_{in} \\ s_{in} \end{Bmatrix} \dot{q}_i , \quad (13)$$

where

$$\begin{Bmatrix} c_{in} \\ s_{in} \end{Bmatrix} = \oint f_i(\sigma) \begin{Bmatrix} \cos n\theta(\sigma) \\ \sin n\theta(\sigma) \end{Bmatrix} d\sigma . \quad (14)$$

The amplitudes q_i may now be treated as generalized coordinates of the motion, so that the kinetic energy becomes a quadratic form in the generalized velocities:

$$T = \frac{1}{2} \sum_i \sum_j m_{ij} \dot{q}_i \dot{q}_j , \quad (15)$$

where

$$m_{ij} = \frac{\rho p^2}{\pi} \sum_{n=1}^{\infty} \frac{c_{in} c_{jn} + s_{in} s_{jn}}{n} . \quad (16)$$

These are the inertia coefficients of the system (cf. Ref. 2, p. 188).

III. Application to a Floating Rectangle

If the curve C is symmetric about the x axis, then in the conformal mapping this axis may be mapped into the real axis of the ζ plane. In such a case, a normal-velocity

distribution which is an odd function of θ , that is, one for which

$$a_n = 0$$

for all n , corresponds to the condition

$$\phi = 0 \quad \text{on} \quad y = 0 .$$

This, however, is just the condition for the surface $y = 0$ to be a free surface. If, consequently, we are dealing with a cylinder floating on the surface of the liquid occupying the half-space $y \leq 0$, the projection of the immersed portion of the cylinder being C_1 , then the problem is equivalent to the previously treated one of the submerged closed cylinder, provided C is the union of C_1 and of its reflection, and the normal-velocity distribution is antisymmetric about the x axis.

The only possible modes of motion are, therefore, those for which the coefficients a_n vanish. Furthermore, since the kinetic energy of the half-space is half of that of the equivalent full space, the inertia coefficients for the floating case are half of those given by Equation (16).

We are interested here in the motion of a rectangular box of beam $2a$ and draft b . The equivalent problem is that of the closed rectangle given by

$$\begin{aligned} -a &\leq x \leq a , \\ -b &\leq y \leq b . \end{aligned}$$

The six degrees of freedom which are described in the Introduction and illustrated in Figure 1 are conveniently grouped into, first, the three which are antisymmetric, and secondly, the three which are symmetric about the y axis. Consequently,

$$\begin{aligned}
 s_{i,2m} &= 4 \int_0^{\frac{1}{4}} f_i(\sigma) \sin 2m\theta(\sigma) d\sigma, \quad m = 1, 2, \dots & i=1, 2, 3 \\
 s_{i,2m+1} &= 0, \quad m = 0, 1, 2, \dots \\
 s_{i,2m} &= 0, \quad m = 1, 2, \dots \\
 s_{i,2m+1} &= 4 \int_0^{\frac{1}{4}} f_i(\sigma) \sin(2m+1)\theta(\sigma) d\sigma, \quad m = 0, 1, \dots & i=4, 5, 6
 \end{aligned}$$

It remains, then, only to obtain the mapping function $\theta(\sigma)$. The appropriate mapping is given by the following special case of the Schwarz-Christoffel transformation:

$$\frac{dz}{d\zeta} = R(1 - 2\zeta^{-2} \cos 2\alpha + \zeta^{-4})^{\frac{1}{2}} \quad (17)$$

where R is real, and the points $\zeta = \pm \exp(\pm i\alpha)$ correspond to the corners of the rectangle. In accordance with Equation (9) we have

$$\frac{ds}{d\theta} = R | 2 \cos 2\alpha - 2 \cos 2\theta |^{\frac{1}{2}}. \quad (18)$$

Equation (18) can be integrated with the aid of elliptic integrals. Using the notation of Ref. 3, we let

$$k = \sin \alpha , k' = \cos \alpha .$$

The coordinates of the rectangle are related to θ by the equations

$$\begin{aligned} x &= \pm a, \\ y &= 2R[E(k, \phi) - k'^2 F(k, \phi)], \\ x &= 2R[E(k', \phi') - k^2 F(k', \phi')], \\ y &= \pm b, \end{aligned} \tag{19}$$

where

$$\phi = \sin^{-1}(\sin \theta/k), \phi' = \sin^{-1}(\cos \theta/k') .$$

In particular, the dimensions of the rectangle are related to R and α by

$$\begin{aligned} a &= 2Rk'^2 B', \\ b &= 2Rk^2 B , \end{aligned}$$

so that the beam-draft ratio $2a/b$ is a function of α only, as is the scale factor γ , defined by

$$\gamma = \frac{2R}{a+b} .$$

Consequently, both α and γ are functions of $2a/b$. For the two ratios of 2 and 3, the values of α and γ , obtained by interpolation from the tables of Ref. 4, are given in Table 1.

Since for the closed rectangle

$$p = 4(a + b) ,$$

the relation between σ and θ is given by

$$\frac{d\sigma}{d\theta} = \frac{\gamma}{8} | 2 \cos 2\alpha - 2 \cos 2\theta |^{\frac{1}{2}} . \quad (20)$$

Though Equation (20) is, of course, integrable in terms of elliptic integrals, there remains the task of inverse interpolation in order to obtain θ as a function of σ . If this task is to be performed by a digital computer, then it is far simpler to include the integration of Equation (20) in the same program, as well as the calculation of $\cos n\theta$ and $\sin n\theta$. The results of these computations are shown in Tables 2 to 5. The inertia coefficients for the six aforementioned degrees of freedom were calculated by numerical integration and summation up to $n = 10$, and are tabulated in matrix form in Table 6.

REFERENCES

1. R. P. Shaw and M. B. Friedman, "Diffraction of Pulses by Arbitrary Two-Dimensional Free Surfaces" (Columbia University, Office of Naval Research Project NR 064-428, Contract Nonr-266(08), Technical Report No. 29, 1961).
2. H. Lamb, Hydrodynamics (6th ed., Dover Publications, New York, 1945).
3. E. Jahnke and F. Emde, Tables of Functions with Formulas and Curves (4th ed., Dover Publications, New York, 1945), p. 52 ff.
4. P. F. Byrd and M. D. Friedman, Handbook of Elliptic Integrals for Engineers and Physicists (Springer-Verlag, Berlin-Göttingen-Heidelberg, 1954), pp. 322-323.

APPENDIX

For the two translatory degrees of freedom, that is $i=1$ and $i=4$ (see Figure 1), the integrals for s_{in} are expressible in closed form.

Consider, first, $i=1$. We have

$$s_{1,2m} = -\frac{\gamma}{2} \int_0^\alpha \sin 2m\theta (2 \cos 2\theta - 2 \cos 2\alpha)^{\frac{1}{2}} d\theta \quad (A1)$$

but the sine of an even multiple of θ can be written as

$$\sin 2m\theta = 2 \sin \theta \cos \theta \sum_{k=0}^{m-1} \frac{(-4)^k (m+k)!}{(2k+1)!(m-k-1)!} \sin^{2k}\theta. \quad (A2)$$

The change of variable

$$\sin \theta = \sin \alpha \sin \psi$$

yields

$$\begin{aligned} s_{1,2m} &= -2\gamma \sin^3 \alpha \sum_{k=0}^{m-1} \frac{(-4)^k (m+k)! \sin^{2k} \alpha}{(2k+1)!(m-k-1)!} \int_0^{\pi/2} \cos^2 \psi \sin^{2k+1} \psi d\psi \\ &= -4\gamma \sin^3 \alpha \sum_{k=0}^{m-1} \frac{(-16)^k (m+k)! k!(k+1)! \sin^{2k} \alpha}{(m-k-1)!(2k+1)!(2k+3)!} . \end{aligned} \quad (A3)$$

Similarly, for $i=4$,

$$s_{4,2m+1} = \frac{\gamma}{2} \int_\alpha^{\pi/2} \sin(2m+1)\theta (2 \cos 2\alpha - 2 \cos 2\theta)^{\frac{1}{2}} d\theta, \quad (A4)$$

but

$$\sin(2m+1)\theta = (-1)^m \sin \theta \sum_{k=0}^m \frac{(-4)^k (m+k)!}{(2k)!(m-k)!} \cos^{2k}\theta. \quad (A5)$$

On changing the variable

$$\cos \theta = \cos \alpha \sin \psi ,$$

we have

$$\begin{aligned} s_{4,2m+1} &= (-1)^m \gamma \cos^2 \alpha \sum_{k=0}^m \frac{(-4)^k (m+k) \cos^2 k \alpha}{(2k) (m-1)} \int_0^{\pi/2} \cos^2 \psi \sin^{2k} \psi d\psi \\ &= (-1)^m \frac{\pi \gamma}{4} \cos^2 \alpha \sum_{k=0}^m \frac{(-1)^k (m+k)! \cos^2 k \alpha}{k! (k+1)! (m-1)!} . \end{aligned} \quad (A6)$$

In the particular case $a/b = 1$, for which

$$\cos^2 \alpha = 1/2 ,$$

we have

$$\begin{aligned} \frac{8}{\pi \gamma} s_{4n} &= 1, & n &= 1 \\ 2(-1)^r \frac{4^{-r} (2r-2)!}{r! (r-1)!} , & n = 4r-1, r = 1, 2, \dots & (A7) \\ 0 & \text{all other } n . \end{aligned}$$

From the closed-form expressions (A3) and (A6), the inertia coefficients m_{11} and m_{44} can be computed to any desired accuracy, for we can find, with the aid of Parseval's theorem, an upper bound to the truncation error due to summing a finite number of terms on the right-hand side of Equation (16). Defining

$$m_{ii}^{(N)} = \frac{\rho p^2}{\pi} \sum_{n=1}^N \frac{c_{in}^2 + s_{in}^2}{n},$$

we have

$$\Delta m_{ii}^{(N)} = m_{ii} - m_{ii}^{(N)} = \frac{\rho p^2}{\pi} \sum_{n=N+1}^{\infty} \frac{c_{in}^2 + s_{in}^2}{n}.$$

Hence

$$\Delta m_{ii}^{(N)} \leq \frac{\rho p^2}{\pi(N+1)} \sum_{n=N+1}^{\infty} (c_{in}^2 + s_{in}^2). \quad (A8)$$

but

$$\frac{1}{\pi} \sum_{n=1}^{\infty} (c_{in}^2 + s_{in}^2) = \oint [f_i(\sigma)]^2 \frac{d\sigma}{d\theta} d\sigma. \quad (A9)$$

Calling the integral on the right-hand side of Equation (A9)

K_i , we have

$$\Delta m_{ii}^{(N)} \leq \frac{\rho p^2}{N+1} [K_i - \frac{1}{\pi} \sum_{n=1}^N (c_{in}^2 + s_{in}^2)]. \quad (10)$$

In particular, for the cases under consideration

$$\begin{aligned} K_1 &= 4(\gamma/8)^2 \int_0^{\alpha} (2 \cos 2\theta - 2 \cos 2\alpha) d\theta \\ &= (\gamma/4)^2 (\sin 2\alpha - 2\alpha \cos 2\alpha), \end{aligned}$$

and

$$\begin{aligned} K_4 &= 4(\gamma/8)^2 \int_{\alpha}^{\pi/2} (2 \cos 2\alpha - 2 \cos 2\theta) d\theta \\ &= (\gamma/4)^2 [\sin 2\alpha + 2(\frac{\pi}{2} - \alpha) \cos 2\alpha]. \end{aligned}$$

TABLE 1

Mapping Parameters

$2a/b$	α	γ
2	0.785398 ($=\pi/4$)	1.180341
3	0.693231	1.175372

TABLE 2

Mapping Function $\theta(\sigma)$, and $\cos n\theta$
for $2a/b = 2$

4σ	θ	$\cos \theta$	$\cos 2\theta$	$\cos 3\theta$	$\cos 4\theta$	$\cos 5\theta$	$\cos 6\theta$	$\cos 7\theta$	$\cos 8\theta$	$\cos 9\theta$	$\cos 10\theta$
.0000	.0000	1.0000	1.0000	1.0000	1.0000	1.0000	1.0000	1.0000	1.0000	1.0000	1.0000
.0500	.0600	.9982	.9928	.9839	.9714	.9554	.9359	.9131	.8871	.8578	.8255
.1000	.1204	.9928	.9711	.9355	.8863	.8242	.7502	.6654	.5709	.4682	.3587
.1500	.1817	.9835	.9347	.8550	.7472	.6148	.4622	.2943	.1167	-.0647	-.2440
.2000	.2445	.9703	.8828	.7428	.5586	.3411	.1034	-.1405	-.3760	-.5891	-.7672
.2500	.3095	.9525	.8144	.5990	.3266	.0232	-.2825	-.5612	-.7867	-.9373	-.9989
.3000	.3777	.9295	.7280	.4239	.0600	-.3124	-.6407	-.8787	-.9228	-.9670	-.8048
.3500	.4506	.9002	.6207	.2172	-.2296	-.6305	-.9056	-.9999	-.8946	-.6107	-.2049
.4000	.5311	.8623	.4870	-.0224	-.5256	-.8841	-.9990	-.8387	-.4474	.0671	.5632
.4500	.6258	.8105	.3138	-.3018	-.8030	-.9999	-.8179	-.3259	.2896	.7954	.9997
.5000	.7854	.7071	.0000	-.7071	-1.0000	-.7071	-.0000	.7071	1.0000	.7071	.0000
.5500	.9450	.5857	-.3138	-.9534	-.8030	.0127	.8719	.9454	.2896	-.6061	-.9997
.6000	1.0397	.5065	-.4870	-.9997	-.5256	.4673	.9990	.5446	-.4474	-.9977	-.5632
.6500	1.1202	.4355	-.6207	-.9761	-.2296	.7762	.9056	.0126	-.8946	-.7919	.2049
.7000	1.1931	.3688	-.7280	-.9057	.0600	.9500	.6407	-.4774	-.9228	-.2548	.8048
.7500	1.2613	.3046	-.8144	-.8008	.3266	.9997	.2825	-.8277	-.7867	.3484	.9989
.8000	1.3263	.2421	-.8828	-.6695	.5586	.9400	-.1034	-.9901	-.3760	.8080	.7672
.8500	1.3891	.1807	-.9347	-.5186	.7472	.7887	-.4622	-.9557	.1167	.9979	.2440
.9000	1.4504	.1201	-.9711	-.3534	.8863	.5663	-.7502	-.7465	.5709	.8836	-.3587
.9500	1.5108	.0600	-.9928	-.1790	.9714	.2954	-.9359	-.4076	.8871	.5140	-.8255
1.0000	1.5708	.0000	-1.0000	.0000	1.0000	.0000	-1.0000	.0000	1.0000	.0000	-1.0000

TABLE 3
Mapping Function $\theta(\sigma)$, and $\sin n\theta$
for $2a/b = 2$

4σ	θ	$\sin \theta$	$\sin 2\theta$	$\sin 3\theta$	$\sin 4\theta$	$\sin 5\theta$	$\sin 6\theta$	$\sin 7\theta$	$\sin 8\theta$	$\sin 9\theta$	$\sin 10\theta$
.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000
.0500	.0600	.0600	.1200	.1790	.2376	.2954	.3522	.4076	.4616	.5140	.5645
.1000	.1204	.1201	.2385	.3534	.4632	.5663	.6612	.7465	.8210	.8836	.9335
.1500	.1817	.1807	.3555	.5186	.6646	.7887	.8868	.9557	.9932	.9979	.9698
.2000	.2445	.2421	.4698	.6695	.8295	.9400	.9946	.9901	.9266	.8080	.6413
.2500	.3095	.3046	.5803	.8008	.9452	.9997	.9523	.8277	.6174	.3484	.0463
.3000	.3777	.3688	.6856	.9057	.9982	.9500	.7678	.4774	.1197	-.2548	-.5935
.3500	.4506	.4355	.7841	.9761	.9733	.7762	.4241	-.0126	-.4468	-.7919	-.9788
.4000	.5311	.5065	.8734	.9997	.8507	.4673	-.0448	-.5446	-.8943	-.9977	-.8263
.4500	.6258	.5857	.9495	.9534	.5960	.0127	-.5754	-.9454	-.9571	-.6061	-.0254
.5000	.7854	.7071	1.0000	.7071	.0000	-.7071	-1.0000	-.7071	.0000	.7071	1.0000
.5500	.9450	.8105	.9495	.3018	-.5960	-.9999	-.5754	.3259	.9571	.7954	-.0254
.6000	1.0397	.8623	.8734	.0224	-.8507	-.8841	-.0448	.8387	.8943	.0671	-.8263
.6500	1.1202	.9002	.7841	-.2172	-.9733	-.6305	.4241	.9999	.4468	-.6107	-.9788
.7000	1.1931	.9295	.6856	-.4239	-.9982	-.3124	.7678	.8787	-.1197	-.9670	-.5935
.7500	1.2613	.9525	.5803	-.5990	-.9452	.0232	.9523	.5612	-.6174	-.9373	.0463
.8000	1.3263	.9703	.4698	-.7428	-.8295	.3411	.9946	.1405	-.9266	-.5891	.6413
.8500	1.3891	.9835	.3555	-.8550	-.6646	.6148	.8868	-.2943	-.9932	-.0647	.9698
.9000	1.4504	.9928	.2385	-.9355	-.4632	.8242	.6612	-.6654	-.8210	.4682	.9335
.9500	1.5108	.9982	.1200	-.9839	-.2376	.9554	.3522	-.9131	-.4616	.8578	.5645
1.0000	1.5708	1.0000	.0000	-1.0000	.0000	1.0000	.0000	-1.0000	.0000	1.0000	.0000

TABLE 4

Mapping Function $\theta(\sigma)$, and $\cos n\theta$
for $2a/b = 3$

4σ	θ	$\cos \theta$	$\cos 2\theta$	$\cos 3\theta$	$\cos 4\theta$	$\cos 5\theta$	$\cos 6\theta$	$\cos 7\theta$	$\cos 8\theta$	$\cos 9\theta$	$\cos 10\theta$
.0000	.0000	1.0000	1.0000	1.0000	1.0000	1.0000	1.0000	1.0000	1.0000	1.0000	1.0000
.0500	.0667	.9978	.9911	.9801	.9646	.9449	.9210	.8930	.8610	.8252	.7857
.1000	.1341	.9910	.9642	.9201	.8595	.7834	.6933	.5907	.4775	.3557	.2275
.1500	.2032	.9794	.9186	.8200	.6876	.5270	.3446	.1482	-.0544	-.2548	-.4446
.2000	.2749	.9625	.8527	.6788	.4541	.1952	-.0784	-.3460	-.5877	-.7852	-.9238
.2500	.3509	.9391	.7637	.4952	.1664	-.1827	-.5095	-.7742	-.9446	-.9999	-.9333
.3000	.4341	.9072	.6462	.2653	-.1649	-.5644	-.8593	-.9947	-.9456	-.7211	-.3628
.3500	.5313	.8622	.4867	-.0230	-.5263	-.8845	-.9990	-.8380	-.4461	.0688	.5647
.4000	.6932	.7692	.1833	-.4872	-.9328	-.9478	-.5252	.1398	.7403	.9990	.7966
.4500	.8533	.6575	-.1353	-.8355	-.9634	-.4314	.3961	.9522	.8562	.1736	-.6278
.5000	.9473	.5838	-.3182	-.9555	-.7974	.0243	.8258	.9400	.2718	-.6226	-.9988
.5500	1.0267	.5176	-.4641	-.9981	-.5692	.4088	.9925	.6186	-.3520	-.9830	-.6657
.6000	1.0982	.4522	-.5856	-.9833	-.3143	.7022	.9536	.1660	-.8025	-.8966	-.0138
.6500	1.1646	.3951	-.6877	-.9386	-.0540	.8959	.7620	-.2938	-.9942	-.4919	.6054
.7000	1.2275	.3366	-.7734	-.8573	.1963	.9894	.4698	-.6732	-.9230	.0519	.9578
.7500	1.2879	.2792	-.8441	-.7505	.4251	.9878	.1264	-.9172	-.6386	.5607	.9516
.8000	1.3464	.2226	-.9010	-.6235	.6235	.9010	-.2225	-1.0000	-.2226	.9010	.6236
.8500	1.4036	.1664	-.9446	-.4810	.7845	.7421	-.5375	-.9210	.2309	.9978	.1013
.9000	1.4598	.1108	-.9755	-.3268	.9031	.5269	-.7863	-.7011	.6310	.8409	-.4448
.9500	1.5155	.0553	-.9939	-.1653	.9756	.2732	-.9454	-.3778	.9036	.4778	-.8507
1.0000	1.5708	.0000	-1.0000	.0000	1.0000	.0000	-1.0000	.0000	1.0000	.0000	-1.0000

TABLE 5

Mapping Function $\theta(\sigma)$, and $\sin n\theta$
for $2a/b = 3$

4σ	θ	$\sin \theta$	$\sin 2\theta$	$\sin 3\theta$	$\sin 4\theta$	$\sin 5\theta$	$\sin 6\theta$	$\sin 7\theta$	$\sin 8\theta$	$\sin 9\theta$	$\sin 10\theta$
.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000	.0000
.0500	.0667	.0666	.1330	.1987	.2636	.3273	.3896	.4501	.5806	.5648	.6186
.1000	.1341	.1337	.2650	.3916	.5111	.6215	.7207	.8069	.8786	.9346	.9738
.1500	.2032	.2018	.3953	.5724	.7261	.8499	.9387	.9890	.9985	.9670	.8957
.2000	.2749	.2714	.5225	.7343	.8910	.9808	.9969	.9382	.8091	.6192	.3828
.2500	.3509	.3437	.6456	.8688	.9861	.9832	.8605	.6329	.3282	-.0165	-.3592
.3000	.4341	.4206	.7632	.9642	.9863	.8255	.5115	.1026	-.3253	-.6928	-.9319
.3500	.5313	.5066	.8736	.9997	.8503	.4665	-.0459	-.5457	-.8950	-.9976	-.8253
.4000	.6932	.6390	.9831	.8733	.3604	-.3184	-.8510	-.9902	-.6723	-.0441	.6045
.4500	.8533	.7534	.9908	.5495	-.2682	-.9022	-.9182	-.3053	.5167	.9848	.7784
.5000	.9473	.8119	.9480	.2951	-.6034	-.9997	-.5639	.3412	.9624	.7825	-.0486
.5500	1.0267	.8556	.8858	.0614	-.8222	-.9126	-.1226	.7857	.9360	.1834	-.7462
.6000	1.0982	.8904	.8106	-.1523	-.9493	-.7120	.3011	.9861	.5967	-.4429	-.9999
.6500	1.1646	.9186	.7259	-.3449	-.9985	-.4442	.6475	.9559	.1078	-.8707	-.7959
.7000	1.2275	.9416	.6339	-.5149	-.9805	-.1452	.8828	.7395	-.3850	-.9987	-.2873
.7500	1.2879	.9602	.5361	-.6609	-.9051	.1555	.9920	.3983	-.7696	-.8280	.3073
.8000	1.3464	.9749	.4339	-.7818	-.7818	.4339	.9749	.0417	-.9749	-.4339	.7818
.8500	1.4036	.9860	.3283	-.8768	-.6201	.6703	.8433	-.3896	-.9730	.0657	.9949
.9000	1.4598	.9938	.2202	-.9451	-.4295	.8499	.6178	-.7131	-.7758	.5412	.8956
.9500	1.5155	.9985	.1105	-.9862	-.2196	.9620	.3260	-.9259	-.4284	.8785	.5256
1.0000	1.5708	1.0000	.0000	-1.0000	.0000	1.0000	.0000	-1.0000	.0000	1.0000	.0000

TABLE 6
Inertia Matrices
($m' = \rho ab$; $r = (a + b)/\sqrt{3}$; $m_{ji} = m_{ij}$)

$2a/b$	2.0	3.0
$\frac{1}{m'}[m_{ij}]$	$\begin{bmatrix} 0.754 & 0.264r & -0.444 & 0 & 0 & 0 \\ & 0.265r^2 & -0.160r & 0 & 0 & 0 \\ & & 0.271 & 0 & 0 & 0 \\ & & & 2.373 & 1.270 & 0.232 \\ & & & & 0.713 & 0.122 \\ & & & & & 0.288 \end{bmatrix}$	$\begin{bmatrix} 0.510 & 0.079r & -0.296 & 0 & 0 & 0 \\ & 0.382r^2 & -0.058r & 0 & 0 & 0 \\ & & 0.177 & 0 & 0 & 0 \\ & & & 3.344 & 1.817 & 0.202 \\ & & & & 1.046 & 0.098 \\ & & & & & 0.164 \end{bmatrix}$
Exact Values:	0.774	0.525
$\frac{1}{m'} m_{11}$		
$\frac{1}{m'} m_{44}$	2.377	3.349

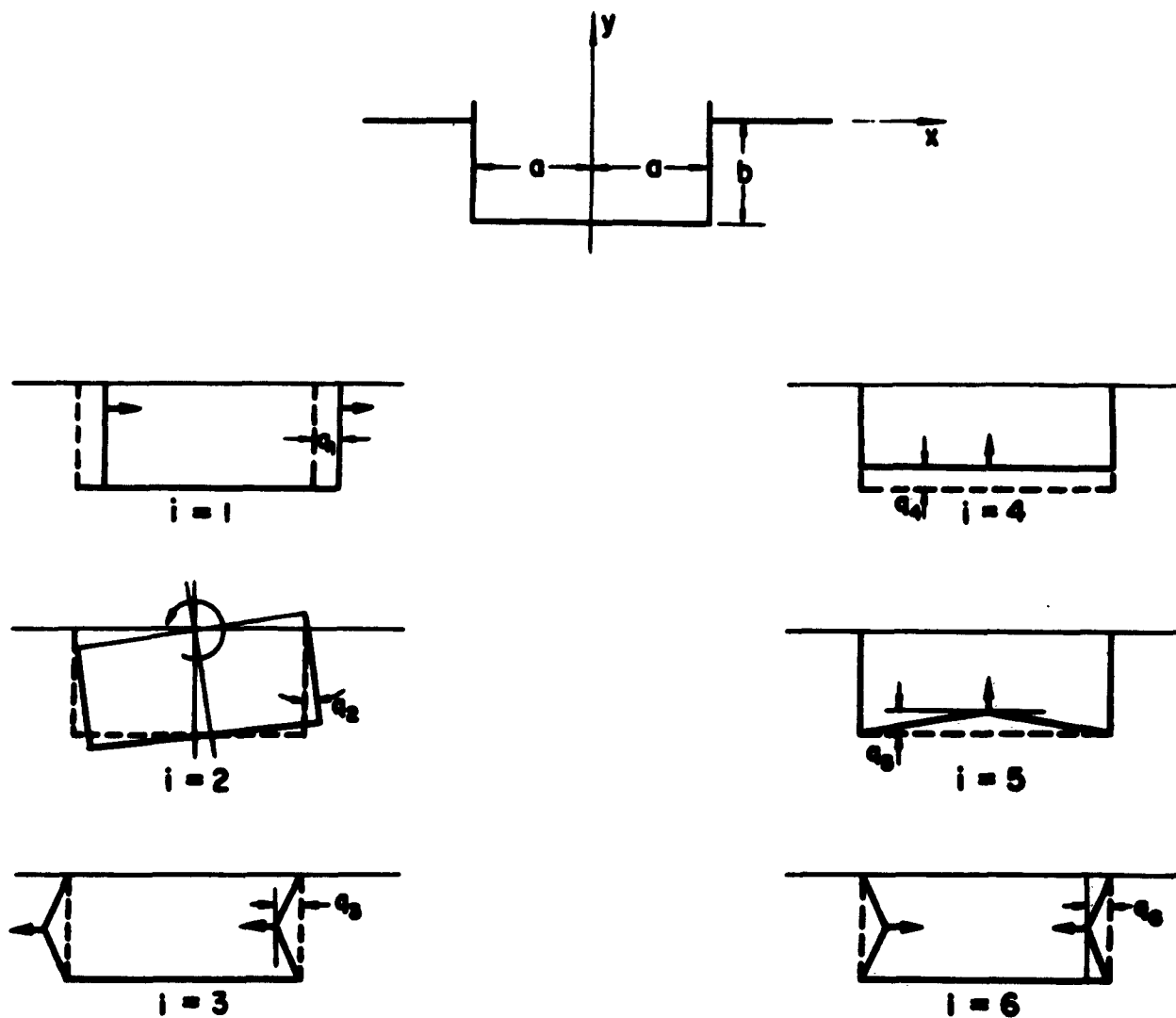


FIG. 1
DISPLACEMENTS CORRESPONDING TO THE
SELECTED DEGREES OF FREEDOM

DISTRIBUTION LIST
for
Contract Nonr 266(08)

Administrative and
Liaison Activities

Chief of Naval Research
Department of the Navy
Washington 25, D. C.
Attn: Code 418 (1)
Code 438 (2)

Office of Naval Research
c/o Hudson Laboratory
148 Palisades Avenue
Dobbs Ferry, New York (1)

Commanding Officer
Office of Naval Research
Branch Office
John Crerar Library Building
86 E. Randolph Street
Chicago 11, Illinois (1)

Commanding Officer
Office of Naval Research
Branch Office
346 Broadway
New York 13, N. Y. (1)

Commanding Officer
Office of Naval Research
Branch Office
1030 E. Green Street
Pasadena, California (1)

Commanding Officer
Office of Naval Research
Navy 100, % Fleet Post Office
New York, N. Y. (2)

Director
Naval Research Laboratory
Washington 25, D. C.
Attn: Tech. Info. Officer (6)
Code 6200 (1)
Code 6205 (1)
Code 6250 (1)
Code 6260 (1)
Code 2029 (1)

Armed Services Technical Information Agency
Arlington Hall Station
Arlington 12, Virginia (10)

Office of Technical Services
Department of Commerce
Washington 25, D. C. (1)

U. S. Department of Defense

Office of the Secretary of Defense
Research and Development Division
The Pentagon
Washington 25, D. C.
Attn: Technical Library (1)

Chief, Defense Atomic Support Agency
The Pentagon
Washington 25, D. C.
Attn: Document Library Branch (2)

Commanding General
Field Command
AFSWP, P. O. Box 5100
Albuquerque, New Mexico (3)

ARMY

Office of the Secretary of the Army
The Pentagon
Washington 25, D. C.
Attn: Army Library (1)

Chief of Staff
Department of the Army
Washington 25, D. C.
Attn: Develop. Branch (R and D Div.) (1)
Research Branch (R and D Div.) (1)
Special Weapons Br. (R and D Div.) (1)

Office of the Chief of Engineers
Asst. Chief of Military Construction
Department of the Army
Bldg. T-7, Gravelly Point
Attn: Library Branch (1)
Structural Branch (Engr. Div.) (1)
Protective Construction Branch
(Pl., Engr., and Contracts) (1)

ARMY (Continued)

Chief, Research and Devel. h
Department of the Army
Washington 25, D. C.
Attn: Special Weapons and Air Defense Div. (1)

Chief of Engineers
Department of the Army
Washington 25, D. C.
Attn: ENGNB (1)

Commanding General
Aberdeen Proving Ground
Maryland
Attn: Director, Ballistic Research Lab. (2)

Commanding Officer, Engineer
Research and Development Lab.
Fort Belvoir, Virginia
Attn: Chief, Tech. Intelligence Br. (1)

Director, Waterways Experiment
Station
P. O. Box 631
Vicksburg, Miss.
Attn: Library (1)

Office of the Chief of Ordnance
Department of the Army
Washington 25, D. C.
Attn: Research and Materials Br. (1)
(Ord. R and D Div.) (1)
ORDTX - AR (1)

Office of the Chief Signal Officer
Department of the Army
Washington 25, D. C.
Attn: Engin. and Tech. Div. (1)

Commanding Officer
Watertown Arsenal
Watertown, Massachusetts
Attn: Laboratory Div. (1)

Commanding Officer
Frankford Arsenal
Bridesburg Station
Philadelphia 37, Pa.
Attn: Laboratory Div. (1)

Office of Ordnance Research
2127 Myrtle Drive
Duke Station
Durham, North Carolina
Attn: Div. of Engin. Sciences (1)

Commanding Officer
Engin. Research Develop. Lab.
Fort Belvoir, Virginia (1)

NAVY

Chief of Naval Operations
Department of the Navy
Washington 25, D. C.
Attn: Op 37 (1)
Op 36 (1)
Op 03EG (1)

Commandant, Marine Corps
Headquarters, U. S. Marine Corps
Washington 25, D. C. (1)

Chief, Bureau of Ships
Department of the Navy
Washington 25, D. C.
Attn: Code 327 (2)
Code 348 (1)
Code 371 (1)
Code 420 (2)
Code 423 (2)
Code 442 (2)
Code 421 (1)

Chief, Bureau of Aeronautics
Department of the Navy
Washington 25, D. C.
Attn: AD-2 (1)
AD-22 (1)
RS-7 (1)
TD-42 (1)

Chief, Bureau of Ordnance
Department of the Navy
Washington 25, D. C.
Attn: Ad3 (1)
Re (1)
Re3 (1)
Re5 (1)
ReN (1)

Director of Naval Intelligence
Navy Department
Washington 25, D. C.
Attn: Op-922V (1)

Chief, Bureau of Yards and Docks
Department of the Navy
Washington 25, D. C.
Attn: Code D-213 (1)
Code D-222 (1)
Code D-410C (1)
Code D-440 (1)
Code D-500 (1)

NAVY (Continued)

Commanding Officer and Director
David Taylor Model Basin
Washington 7, D. C.

Attn: Code 140 (1)
Code 600 (1)
Code 700 (1)
Code 720 (1)
Code 725 (1)
Code 731 (1)
Code 740 (2)

Commander
U. S. Naval Ordnance Laboratory
White Oak, Maryland

Attn: Tech. Library (2)
Tech. Evaluation Dep. (1)
EE (1)
EH (1)
R (1)

Director
Material Laboratory
New York Naval Shipyard
Brooklyn 1, New York (1)

Commanding Officer and Director
U. S. Naval Electronics Lab.
San Diego 52, California
Attn: Code 4223 (1)

Officer-in-Charge
Naval Civil Engin. Research
Naval Civil Engin. Research and
Evaluation Laboratory
U. S. Naval Construction
Battalion Center
Port Hueneme, California
Attn: Code 753 (1)

Director
Naval Air Experimental Station
Naval Air Material Center
Naval Base
Philadelphia 12, Pa
Attn: Materials Laboratory (1)
Structures Lab. (1)

Officer-in-Charge
Underwater Explosion Research Div.
Norfolk Naval Shipyard
Portsmouth, Virginia
Attn: Dr. A.H. Keil (2)

Commander
U.S. Naval Providing Grounds
Dahlgren, Virginia (1)

Superintendent
Naval Gun Factory
Washington 25, D. C. (1)

Commander
Naval Ordnance Test Station
Inyokern, China Lake, California
Attn: Physics Division (1)
Mechanics Branch (1)

Commander
Naval Ordnance Test Station
Underwater Ordnance Division
3202 E. Foothill Boulevard
Pasadena 8, California
Attn: Structures Division (1)

Commanding Officer and Director
Naval Engineering Experiment Station
Annapolis, Maryland (1)

Superintendent
Naval Post Graduate School
Monterey, California (1)

Commandant
Marine Corps Schools
Quantico, Virginia
Attn: Director, Marine Corps
Development Center (1)

AIR FORCE

Commanding General
U. S. Air Force
Washington 25, D. C.
Attn: Research and Development Div. (1)

Commander
Air Materiel Command
Wright-Patterson Air Force Base
Dayton, Ohio
Attn: WCOSI (1)

Commander
U. S. Air Force Institute of Technology
Wright-Patterson Air Force Base
Dayton, Ohio
Attn: Chief, Applied Mech. Group (1)

Director of Intelligence
Headquarters, U. S. Air Force
Washington 25, D. C.
Attn: P. V. Br. (Air Targets Div.) (1)
AFOIN-1B2 (2)

Commander
Air Research and Development
Command
P. O. Box 1395
Attn: RDMPE (1)

Commander
WADD
Wright-Patterson Air Force Base
Dayton, Ohio
Attn: WARC (1)
WARMS (1)
WARMD (1)

Commanding Officer
USNMOCU
Kirtland Air Force Base
Albuquerque, New Mexico
Attn: Code 20 (1)
(Dr. J. N. Brennan) (1)

OTHER GOVERNMENT ACTIVITIES

U. S. Atomic Energy Commission
Washington 25, D. C.
Attn: Director of Research (2)

Director
National Bureau of Standards
Washington 25, D. C.
Attn: Division of Mechanics (1)
Engin. Mechanics Sec. (1)
Aircraft Structures (1)

Commander
U. S. Coast Guard
1300 E. Street, N. W.
Washington 25, D. C.
Attn: Chief, Testing and
Development Div. (1)

U.S. Maritime Administration
General Administration Office Bldg.
Washington 25, D. C.
Attn: Chief, Div. of Preliminary
Design (1)

National Advisory Committee for
Aeronautics
1512 E. Street, N.W.
Washington 25, D. C.
Attn: Loads and Structures Div. (2)

Director
Langley Aeronautical Laboratory
Langley Field, Virginia
Attn: Structures Div. (2)

Director
Forest Products Laboratory
Madison, Wisconsin (1)

Civil Aeronautical Administration
Department of Commerce
Washington 25, D. C.
Attn: Chief, Airframes and Equip.Br. (1)

National Sciences Foundation
1520 H. Street, N. W.
Washington 25, D. C.
Attn: Engin. Science Division (1)

National Academy of Science
2101 Constitution Avenue
Washington 25, D. C.
Attn: Tech. Director, Committee On
Ships' Structural Design (1)
Executive Secretary, Committee
on Underwater Warfare (1)

Director, Operations Research Office
John Hopkins University
7100 Connecticut Avenue
Chevy Chase, Maryland
Washington 15, D. C. (1)

Dr. Alvin C. Graves, Director
J-Division, Los Alamos Scientific Lab.
P. O. Box 1663
Los Alamos, New Mexico (1)

U. S. Atomic Energy Commission
Classified Technical Library
1901 Constitution Avenue, N. W.
Washington, D. C.
Attn: Mrs. Jean M. O'Leary
for Dr. Paul C. Fine (1)

U.S. Atomic Energy Commission
Classified Tech. Library
Tech. Information Service
1901 Constitution Avenue, N.W.
Attn: Mrs. Jean M.O'Leary (1)

Sandia Corporation, Sandia Base
Albuquerque, New Mexico
Attn: Dr. Walter A. MacNair (1)

Legislative Reference Service
Library of Congress
Washington 25, D. C.
Attn: Dr. E. Wenk (1)

U.S. Atomic Energy Commission
1901 Constitution Avenue, N.W.
Washington, D. C. (1)

OTHER GOVERNMENT ACTIVITIES (Continued)

Massachusetts Institute of
Technology
Cambridge 39, Massachusetts
Attn: Dr. Charles H. Norris (1)

INVESTIGATORS ACTIVELY ENGAGED
IN RELATED RESEARCH

Professor Lynn S. Beedle
Fritz Engineering Laboratory
Lehigh University
Bethlehem, Pennsylvania (1)

Professor H. H. Bleich
Department of Civil Engineering
Columbia University
New York 27, New York (1)

Professor R. L. Bisplinghoff
Dept. of Aeronautical Engineering
Massachusetts Institute of Techno.
Cambridge 39, Massachusetts (1)

Professor B. A. Boley
Department of Civil Engineering
Columbia University
New York 27, N.Y.

Professor Eugene J. Brunelle, Jr.
Department of Aeronautical Engin.
Princeton University
Princeton, New Jersey (1)

Professor G.F. Carrier
Pierce Hall
Harvard University
Cambridge 38, Massachusetts (1)

Professor J.E. Cermak
Department of Civil Engin.
Colorado State University
Fort Collins, Colorado (1)

Professor Herbert Deresiewicz
Department of Civil Engin.
Columbia University
632 West 125th Street
New York 27, N.Y. (1)

Professor Lloyd Donnell
Department of Mechanics
Illinois Institute of Technology
Technology Center
Chicago 16, Illinois (1)

Professor D. C. Drucker, Chairman
Division of Engineering
Brown University
Providence 12, Rhode Island (1)

Professor A. C. Eringen
Department of Aeronautical Engineering
Purdue University
Lafayette, Indiana (1)

Professor W. Flugge
Department of Mechanical Engineering
Stanford University
Stanford, California (1)

Mr. Martin Goland
Midwest Research Institute
4049 Pennsylvania Avenue
Kansas City 2, Missouri (1)

Professor J. N. Goodier
Department of Mechanical Engineering
Stanford University
Stanford, California (1)

Professor L.E. Goodman
Engineer Experiment Station
University of Minnesota
Minneapolis, Minnesota (1)

Professor W.J. Hall
Department of Civil Engineering
University of Illinois
Urbana, Illinois (1)

Professor R.P. Harrington, Head
Department of Aeronautical
Engineering
University of Cincinnati
Cincinnati 21, Ohio (1)

Professor M. Hetenyi
The Technological Institute
Northwestern University
Evanston, Illinois (1)

Professor P.G. Hodge
Department of Mechanics
Illinois Institute of Technology
Technology Center
Chicago 16, Illinois (1)

Professor W.J. Hoff
Stanford University
Stanford, California (1)

**INVESTIGATORS ACTIVELY ENGAGED
IN RELATED RESEARCH (Continued)**

Professor W. H. Koppmann, II
Department of Mechanical Engr.
John Hopkins University
Baltimore, Maryland (1)

Professor Bruce G. Johnson
University of Michigan
Ann Arbor, Michigan (1)

Professor J. Kempner
Department of Aeronautical Engr.
and Applied Mechanics
Polytechnic Institute of Brooklyn
99 Livingston Street
Brooklyn 2, New York (1)

Professor H. L. Langhaar
Department of Theoretical
and Applied Mechanics
University of Illinois
Urbana, Illinois (1)

Professor B. J. Lazan, Director
Engineering Experiment Station
University of Minnesota
Minneapolis 14, Minnesota (1)

Professor E. H. Lee
Division of Applied Mathematics
Brown University
Providence 12, Rhode Island (1)

Professor George H. Lee
Director of Research
Rensselaer Polytechnic Institute
Troy, New York

Mr. M. M. Lemcoe
Southwest Research Institute
8500 Culebra Road
San Antonio 6, Texas (1)

Professor Paul Lieber
Geology Department
Rensselaer Polytechnic Institute
Troy, New York (1)

Professor Hsu Lo
School of Engineering
Purdue University
Lafayette, Indiana (1)

Professor R. D. Mindlin
Department of Civil Engineering
Columbia University
New York 27, N. Y. (1)

Dr. A. Nadai
136 Cherry Valley Road
Pittsburgh 21, Pennsylvania (1)

Professor William A. Nash
Department of Engineering Mechanics
University of Florida
Gainesville, Florida (1)

Professor N. M. Newmark
Department of Civil Engineering
University of Illinois
Urbana, Illinois (1)

Professor Aris Phillips
Department of Civil Engineering
15 Prospect Street
Yale University
New Haven, Connecticut (1)

Professor W. Prager, Chairman
Physical Sciences Council
Brown University
Providence 12, Rhode Island (1)

Professor E. Reissner
Department of Mathematics
Massachusetts Institute of Technology
Cambridge 39, Massachusetts (1)

Professor M. A. Sadowsky
Department of Mechanics
Rensselaer Polytechnic Institute
Troy, New York (1)

Dr. B. W. Shaffer
Department of Mechanical Engineering
New York University
45 Fourth Avenue
New York 53, New York (1)

Professor C. B. Smith
College of Arts and Science
Department of Mathematics
Walker Hall
University of Florida
Gainesville, Florida (1)

Professor J. Stallmeyer
Department of Civil Engineering
University of Illinois
Urbana, Illinois (1)

Professor Eli Sternberg
Brown University
Providence 12, Rhode Island (1)

INVESTIGATORS ACTIVELY ENGAGED
IN RELATED RESEARCH (Continued)

Professor S.P. Timoshenko
School of Engineering
Stanford University
Stanford, California (1)

Professor A.A. Vealestos
Department of Civil Engineering
University of Illinois
Urbana, Illinois (1)

Professor Enrico Volterra
Department of Engineering Mech.
University of Texas
Austin 12, Texas (1)

Professor Dana Young
Yale University
New Haven, Connecticut (1)

Project Staff (10)